

# Surface waves in fiber-reinforced anisotropic elastic media

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## 1. Introduction

After the pioneer work of Rayleigh (1885), many investigators have been studied the problem extensively under different conditions. They have contributed in a wide range towards its application in various fields i.e. Seismology, geophysics, acoustics, telecommunications and environmental sciences etc.

In last many investigations the reinforcement effect was neglected. Belfield et al. (1983) has introduced this concept. The components of reinforced composite behave like a single anisotropic component till they are in elastic condition. Both the parts have no relative displacement between them.

Chatopadhyay and Venkateswarlu (1998) studied the stress generated because of moving load in fiber reinforced half space. Sengupta&Nath (2001) used the method of potential to decouple the P & SV motions for the study of surface waves, but it was not justified. Therefore Singh (2004) pointed out that the results of Sengupta and Nath (2001) are in error regarding Rayleigh and Stoneley waves.

The results are compared with corresponding classical results for isotropic elastic medium in the absence of reinforced elastic parameters.

### 1.1 Basic Equation

In a Fiber-reinforced linearly elastic anisotropic media Belfield et al. (1983) formulated the below expression for constitutive equations versus reinforcement direction  $\vec{a}$

$$\sigma_{ij} = \lambda e_{kk} \delta_{ij} + 2\mu_T e_{ij} + \alpha(a_k a_m e_{km} \delta_{ij} + e_{kk} a_j a_i) + 2(\mu_L - \mu_T)(a_i a_k e_{kj} + a_j a_k e_{ki}) + \beta a_k a_m e_{km} a_i a_j, \dots \dots \dots (1.1)$$

Here the constant and the variable components mentioned in this above equations are as:

1.  $\sigma_{ij}$  stands for stress and
2.  $e_{ij}$  stands for strain and
3.  $\lambda, \mu_T$  refers to the elastic constants and
4.  $\alpha, \beta, (\mu_L - \mu_T)$  refers to the reinforcement parameters

$$\vec{a} = (a_1, a_2, a_3), \quad a_1^2 + a_2^2 + a_3^2 = 1$$

The direction of the Fiber is assumed as  $\vec{a} = (1, 0, \text{and } 0)$ .

The below expression shows the strains in the form of displacements  $u_i$

$$e_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \dots \dots \dots (1.2)$$

Equation (1.1) then yields

$$\begin{aligned} \sigma_{11} &= A_{11} \frac{\partial u_1}{\partial x_1} + A_{12} \frac{\partial u_2}{\partial x_2} + A_{13} \frac{\partial u_3}{\partial x_3}, \\ \sigma_{22} &= A_{12} \frac{\partial u_1}{\partial x_1} + A_{22} \frac{\partial u_2}{\partial x_2} + A_{23} \frac{\partial u_3}{\partial x_3}, \\ \sigma_{33} &= A_{12} \frac{\partial u_1}{\partial x_1} + A_{23} \frac{\partial u_2}{\partial x_2} + A_{22} \frac{\partial u_3}{\partial x_3}, \\ \sigma_{21} = \sigma_{12} &= \mu_L \left( \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right), \\ \sigma_{31} = \sigma_{13} &= \mu_L \left( \frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1} \right), \\ \sigma_{23} = \sigma_{32} &= \mu_T \left( \frac{\partial u_2}{\partial x_3} + \frac{\partial u_3}{\partial x_2} \right) \dots \dots \dots (1.3) \end{aligned}$$

Here

$$A_{11} = \lambda + 2\alpha + 4\mu_T - 2\mu_T + \beta, \quad A_{12} = \lambda + \alpha, \dots \dots \dots (1.4)$$

$$A_{22} = \lambda + 2\mu_T, \quad A_{23} = \lambda,$$

Neglecting the body forces, the motion's equation can be mentioned as

$$\frac{\partial \sigma_{ij}}{\partial x_j} = \rho \frac{\partial^2 u_i}{\partial t^2} \quad (i, j = 1, 2, 3) \dots \dots \dots (1.5)$$

Here  $\rho$  stands for density of the elastic medium,

With (1.3) in (1.5)

$$\begin{aligned} A_{11} \frac{\partial^2 u_1}{\partial x_1^2} + (A_{12} + \mu_L) \frac{\partial^2 u_2}{\partial x_1 \partial x_2} + (A_{12} + \mu_L) \frac{\partial^2 u_3}{\partial x_1 \partial x_3} + \mu_L \frac{\partial^2 u_1}{\partial x_2^2} + \mu_L \frac{\partial^2 u_1}{\partial x_3^2} &= \rho \frac{\partial^2 u_1}{\partial t^2}, \\ A_{22} \frac{\partial^2 u_2}{\partial x_2^2} + (A_{12} + \mu_L) \frac{\partial^2 u_1}{\partial x_1 \partial x_2} + (A_{23} + \mu_T) \frac{\partial^2 u_3}{\partial x_1 \partial x_2} + \mu_L \frac{\partial^2 u_2}{\partial x_1^2} + \mu_L \frac{\partial^2 u_2}{\partial x_3^2} &= \rho \frac{\partial^2 u_2}{\partial t^2} \end{aligned}$$

$$A_{22} \frac{\partial^2 u_3}{\partial x_3^2} + (A_{23} + \mu_T) \frac{\partial^2 u_2}{\partial x_2 \partial x_3} + (A_{12} + \mu_L) \frac{\partial^2 u_1}{\partial x_1 \partial x_3} + \mu_L \frac{\partial^2 u_3}{\partial x_1^2} + \mu_T \frac{\partial^2 u_3}{\partial x_2^2} = \rho \frac{\partial^2 u_3}{\partial t^2}, \dots\dots\dots (1.6)$$

When putting  $\frac{\partial}{\partial x_3} = 0, u_3 = 0$  in first two equations of (1.6) we get deformation in the  $x_1x_2$ -plane as shown below

$$A_{11} \frac{\partial^2 u}{\partial x^2} + B_2 \frac{\partial^2 v}{\partial x \partial y} + B_1 \frac{\partial^2 u}{\partial y^2} = \rho \frac{\partial^2 u}{\partial t^2},$$

$$A_{22} \frac{\partial^2 v}{\partial y^2} + B_2 \frac{\partial^2 u}{\partial x \partial y} + B_1 \frac{\partial^2 v}{\partial x^2} = \rho \frac{\partial^2 v}{\partial t^2}. \dots\dots\dots (1.6 a)$$

The third equation of (1.6) is identically satisfied & we used the symbolizations as

$$X_1 = x, X_2 = y, u_1 = u, u_2 = v, B_1 = \mu_L, B_2 = A_{12} + \mu_L.$$

**1.2 Plane Wave in Infinite Media**

We want to investigate the behavior of plane waves in  $xy$ -plane with displacements  $u$  in the direction of  $x$ -axis while displacement  $v$  in the direction of  $y$ -axis.

Here in the below figure the component of the plane wave are as:

1.  $\omega$  is the circular frequency
2.  $\theta$  is angle with the direction of plane wave transmission with  $y$ -axis

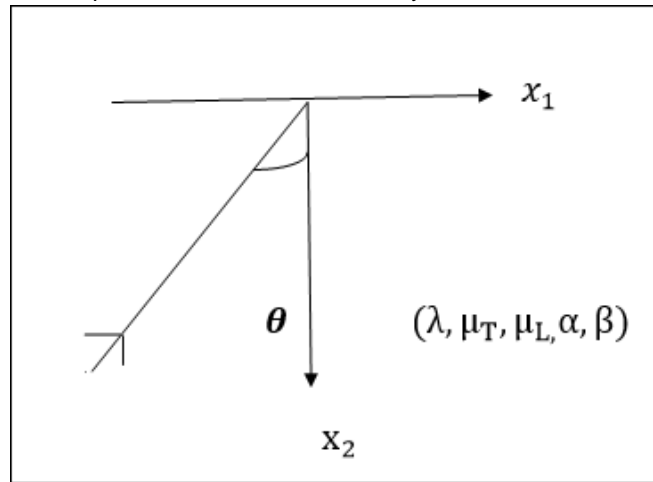


Fig 1.1

$$u = U \exp(ip_1), \quad v = V \exp(ip_1), \dots\dots\dots (1.7)$$

Substituting equation (1.7) in equation (1.1a), we have as in Singh and Singh(2004)

$$-(D_1 - \rho C^2)U + B_2 \sin\theta \cos\theta V = 0,$$

$$B_2 \sin\theta \cos\theta U - (D_2 - \rho C^2)V = 0. \dots\dots\dots (1.8)$$

$D_1$  and  $D_2$  are given by

$$D_1(\theta) = A_{11} \sin^2\theta + B_1 \cos^2\theta,$$

$$D_2(\theta) = A_{22} \cos^2\theta + B_1 \sin^2\theta \dots\dots\dots (1.9)$$

The set of homogeneous equations (1.8) in  $U$  and  $V$  has a non-trivial solution only if

$$|-(D_1 - \rho C^2)B_2 \sin\theta \cos\theta B_2 \sin\theta \cos\theta - (D_2 - \rho C^2)|, \dots\dots\dots (1.10)$$

Outcomes of equation (1.10)

$$2\rho C^2(\theta) = (D_1 + D_2) \pm [(D_1 - D_2)^2 + 4B_2^2 \sin^2\theta \cos^2\theta]^{\frac{1}{2}} \dots\dots\dots (1.11)$$

In equation (1.11) let's suppose  $C_1(\theta)$  and  $C_2(\theta)$  represents  $C$  in connection with above symbols.

Here  $c_1$  is the velocity of the quasi-P waves and  $c_2$  is the velocity of the quasi-S waves.

**1.3 Propagation of Rayleigh Waves**

In a Fiber reinforced anisotropic half-space the surface waves are transmitting in the direction of  $X$ -axis. Below are the components of Rayleigh waves:

1.  $\omega$  refers to the circular frequency,
2.  $K$  refers to the wave number and
3.  $C_R$  refers to the phase velocity

We may assume result of (1.6a) as

$$u = U e^{-kqy} e^{ik(C_R t - x)},$$

$$v = V e^{-kqy} e^{ik(C_R t - x)}, \dots\dots\dots (1.12)$$

$q$  is assumed to be real and positive. Putting the values of the displacements in (1.6a), we get

$$(\rho C_R^2 - A_{11} + B_1 q^2)U + iq B_2 V = 0,$$

$$iqB_2U + (\rho C_R^2 - B_1 + A_{11}q^2)V = 0, \dots\dots\dots (1.13)$$

The values of q may be obtained from

$$|(\rho C_R^2 - A_{11} + B_1q^2)iqB_2iqB_2\rho C_R^2 - B_1 + A_{22}q^2| = 0,$$

Which on simplification we get

$$q_1^2, q_2^2 = \frac{-\{(A_{22}+B_1)\rho C_R^2 - A_{11}A_{22} - B_1^2 + B_2^2\} \pm \left[ \{(A_{22}+B_1)\rho C_R^2 - A_{11}A_{22} - B_1^2 + B_2^2\}^2 - 4B_1A_{22}(\rho C_R^2 - A_{11})(\rho C_R^2 - B_1) \right]^{1/2}}{2B_1A_{22}} \dots\dots(1.14)$$

Therefore the solution (1.12) can be written as

$$u = (U_{11}e^{-q_1ky} + U_{12}e^{-q_2ky})e^{-ik(C_Rt-x)},$$

$$v = (V_{21}e^{-q_1ky} + V_{22}e^{-q_2ky})e^{-ik(C_Rt-x)}, \dots\dots\dots (1.15)$$

(U<sub>11</sub>, U<sub>12</sub>) and (V<sub>21</sub>, V<sub>22</sub>) are not independent but are connected by (1.13) for q=q<sub>1</sub> and q<sub>2</sub>.

Taking second number of (1.13), we get

$$\frac{U_{11}}{V_{21}} = m_1, \frac{U_{21}}{V_{22}} = m_2, \dots\dots\dots(1.16)$$

where m<sub>1</sub> = iM<sub>1</sub>, m<sub>2</sub> = iM<sub>2</sub>.....(1.17)

$$M_1 = \frac{\rho C_R^2 - B_1 + A_{22}q_1^2}{q_1B_2}, \dots\dots\dots(1.18)$$

and

$$M_2 = \frac{\rho C_R^2 - B_1 + A_{22}q_2^2}{q_2B_2}, \dots\dots\dots (1.19)$$

where q<sub>1</sub> and q<sub>2</sub> are real and positive quantities defined in (1.14)

Therefore (1.15) becomes

$$u = (m_1V_{21}e^{-q_1ky} + m_2V_{22}e^{-q_2ky})e^{ik(C_Rt-x)},$$

$$v = (V_{21}e^{-q_1ky} + V_{22}e^{-q_2ky})e^{ik(C_Rt-x)}. \dots\dots\dots(1.20)$$

**Boundary Conditions**

The displacement in (1.20) must be true at the borderlinesituations,

$$\sigma_{21} = \sigma_{22} = 0, \text{ at } y = 0, \dots\dots\dots (1.21)$$

Where  $\sigma_{21}$  and  $\sigma_{22}$  are defined in (1.3). For plane deformation in the x<sub>1</sub>x<sub>2</sub>-plane,  $\frac{\partial}{\partial x_3} = 0, u_3 = 0$ . Equation (1.3) then yields

$$\sigma_{21} = \mu_L \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right),$$

$$\sigma_{22} = A_{12} \frac{\partial u}{\partial x} + A_{22} \frac{\partial v}{\partial y}, \dots\dots\dots(1.22)$$

He we denote them as x<sub>1</sub> = x, x<sub>2</sub> = y, u<sub>1</sub> = u and u<sub>2</sub> = v.

From equations (1.21) and (1.22), we get

$$(m_1q_1 + i)V_{21} + (m_2q_2 + i)V_{22} = 0,$$

$$(A_{12}m_1i + A_{22}q_1)V_{21} + m_2A_{12}i + A_2q_2)V_{22} = 0. \dots\dots\dots(1.23)$$

Eliminating V<sub>21</sub> and V<sub>22</sub> from (5.3), we get

$$|m_1q_1 + im_2q_2 + iA_{12}m_1i + A_{22}q_1m_2A_{12}i + A_2q_2| = 0. \dots\dots\dots(1.24)$$

On simplification, equation (1.24) becomes

$$[(m_1q_1 + i)(m_2A_{12}i + A_{22}q_2) - (m_2q_2 + i)(A_{12}m_1i + A_{22}q_1)] = 0. \dots\dots\dots (1.25)$$

Here in a Fiber-reinforced elastic material this above equation is referred as equation of velocity of Rayleigh waves.

**Particular Case**

If we put  $\alpha, \beta$  and  $|\mu_L - \mu_T|$  equal to zero, the elastic coefficients become

$$A_{11} = \lambda + 2\mu_T, A_{12} = \lambda, B_1 = \mu_T,$$

$$A_{22} = \lambda + 2\mu_T, A_{23} = \lambda, B_2 = \lambda + \mu_T. \dots\dots\dots (1.21)$$

From equations (1.18) and (1.19) we get

$$M_1 = q_1, M_2 = \frac{1}{q_2} \text{ then}$$

$$q_1^2 = \left(1 - \frac{C_R^2}{\beta^2}\right), q_2^2 = \left(1 - \frac{C_R^2}{\alpha^2}\right), \dots\dots\dots (1.27)$$

Where

$$\alpha^2 = \frac{\lambda + 2\mu_T}{\rho} \text{ And } \beta^2 = \frac{\mu_T}{\rho}. \dots\dots\dots (1.28)$$

$\lambda, \mu_T$  are Lamé constants. Using equations (1.21)-(1.28), the velocity equation (1.25) reduces to

$$\left(\frac{C_R^2}{\beta^2} - 2\right)^2 = 4\left(1 - \frac{C_R^2}{\alpha^2}\right)^2 \left(1 - \frac{C_R^2}{\beta^2}\right)^2. \dots\dots\dots (1.29)$$

**1.4 Transmission of Love Waves**

Consider an isotropic layer of Fiber-reinforced elastic medium of thickness H over a homogeneous anisotropic Fiber-reinforced elastic half space.

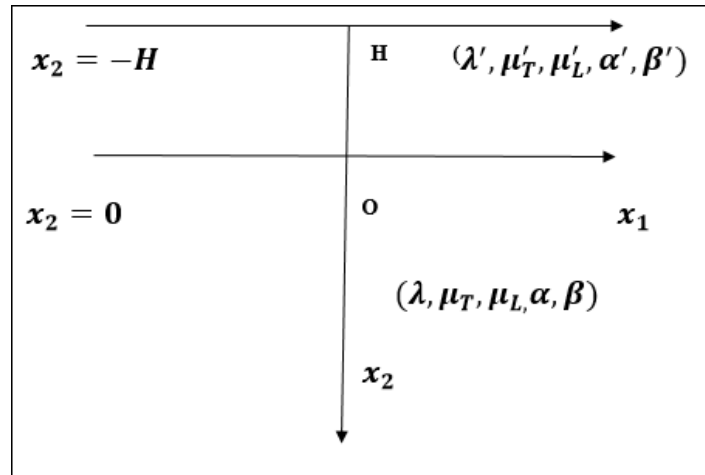


Fig. 1.2

Generally it is presumed that the wave is transmitting towards the direction of plane  $x_3$ . For antiplane strain in the upper layer motion's expression for Love waves can be derived from third equation of (1.1) by putting  $u_1 = u_2 = 0, u_3 = w, x_1 = x, x_2 = y, x_3 = z$  and  $\frac{\partial}{\partial x_3} = 0$ , as

$$\mu'_L \frac{\partial^2 w'}{\partial x^2} + \mu'_T \frac{\partial^2 w'}{\partial y^2} = \rho' \frac{\partial^2 w'}{\partial t^2}, \tag{1.30}$$

where  $\rho'$  is density and  $(\mu'_L, \mu'_T)$  are reinforced anisotropic elastic parameters respectively for the layer. The equation of motion for the half space is

$$\mu_L \frac{\partial^2 w}{\partial x^2} + \mu_L \frac{\partial^2 w}{\partial y^2} = \rho \frac{\partial^2 w}{\partial t^2}, \tag{1.31}$$

Here in a half space the corresponding quantities are  $\mu_L, \mu_T$  and  $\rho$

**1.5 Solution of the Equation of the Motion**

When a plane harmonic wave is transmitting in the upper layer in the direction of  $x_3$  plane we can presume the resolution of equation (1.30) expressed as

$$w' = W' e^{i\omega(t - \frac{x}{c_L}) - p'y}, \tag{1.32}$$

where  $W'$  is the amplitude factor,  $\omega$  and  $c_L$  are circular frequency and phase velocity respectively in the layer. Substituting equation (1.32) in equation (1.30) we get

$$\frac{\omega^2}{c_L^2} \mu'_L - \mu'_T p'^2 = \rho' \omega^2. \tag{1.33}$$

From equation (1.33), we can mention  $p'$  as  $p' = \pm ip'_1$ , where

$$p'_1 = \frac{\omega}{c_L} \sqrt{\frac{\rho' c_L^2 - \mu'_L}{\mu'_T}} = K \sqrt{\frac{\rho' c_L^2 - \mu'_L}{\mu'_T}}. \tag{1.34}$$

Therefore equation (1.32) can be written as

$$w'(x, y, t) = W'_1 e^{i\omega(t - \frac{x}{c_L}) - ip'_1 y} + W'_2 e^{i\omega(t - \frac{x}{c_L}) + ip'_1 y}. \tag{1.35}$$

Similarly the solution of equation (1.32) for half-space is

$$w(x, y, t) = W_1 e^{i\omega(t - \frac{x}{c_L}) - ip_1 y} + W_2 e^{i\omega(t - \frac{x}{c_L}) + ip_1 y}, \tag{1.36}$$

where

$$p_1^2 = K^2 \left( \frac{\rho c_L^2 - \mu_L}{\mu_T} \right) \tag{1.37}$$

$$w(x, y, t) \rightarrow 0 \text{ as } y \rightarrow \infty.$$

Therefore  $p_1$  should be purely imaginary.

Let

$$p_1 = ip_2 = i \left( K \sqrt{\frac{\mu_L - \rho c_L^2}{\mu_T}} \right). \tag{1.38}$$

Equation (1.36) then show that

$$W_1 = 0. \tag{1.39}$$

**Boundary Conditions**

The below mentioned values of the components must be satisfied.

$$w'_3 = w_3 aty = 0, \dots\dots\dots (1.40)$$

$$\tau_{23} \text{ is defined in (1.3) and can be written as (putting } u_2 = u_1 = 0, u_3 = w, x_1 = x, x_2 = y, x_3 = z \text{ and } \frac{\partial}{\partial x_3} = 0$$

$$\tau_{23} = \mu_T \frac{\partial w}{\partial y};$$

$$\text{Similarly } \tau'_{23} = \mu'_T \frac{\partial w'}{\partial y}. \dots\dots\dots (1.41)$$

From equations (1.31) – (1.41), we get

$$W'_1 e^{ip_1 h} - W'_2 e^{-ip_1 h} = 0,$$

$$W'_1 + W'_2 - W_2 = 0, \dots\dots\dots (1.42)$$

$\mu'_T p_1 (W'_1 - W'_2) + \mu_T p_1 W_2 = 0.$   
 Eliminating  $W'_1, W'_2$  and  $W_2$  from equation (1.42) we obtain

$$i \tan(p_1 H) = \frac{\mu_T p_1}{p_1 \mu'_T}. \dots\dots\dots (1.43)$$

Substituting for  $p_1$  and  $p_1$  from (1.35) and (1.39) in (1.43), we get

$$\tan \left[ \left( \frac{\rho' c_{L'}^2 - \mu_L}{\mu_T} \right) KH \right] = \frac{(\mu_L - \rho c_{L'}^2)^{\frac{1}{2}} \cdot \mu_T}{\left( \frac{\rho' c_{L'}^2 - \mu_L}{\mu_T} \right)^{\frac{1}{2}} \cdot \mu_T}. \dots\dots\dots (1.44)$$

In an anisotropic Fiber-reinforced elastic medium the above equation (1.44) refer to the Love wave's frequency equation.

**Particular Case**

If we put  $\mu_L = \mu_T$ , in (1.44), we get

$$\tan \left[ \left( \frac{c_L^2}{\beta^2} - 1 \right)^{\frac{1}{2}} KH \right] = \frac{\mu_T}{\mu'_T} \cdot \frac{(1 - \frac{c_L^2}{\beta^2})^{\frac{1}{2}}}{(\frac{c_L^2}{\beta^2})^{\frac{1}{2}}}, \dots\dots\dots (1.45)$$

where  $\beta'^2 = \frac{\mu'_T}{\rho'}$  and  $\beta^2 = \frac{\mu_T}{\rho}$ ,

In a homogeneous elastic layer this above equation refers to the classical frequency of Love waves over a isotropic half space.

**1.6 Propagation of Stoneley Waves**

When Rayleigh waves are transmitting over the common boundary of two fiber-reinforced homogeneous elastic half spaces say  $M'$  and  $M''$  we can generate Stoneley waves as shown in (Fig. 1.3)

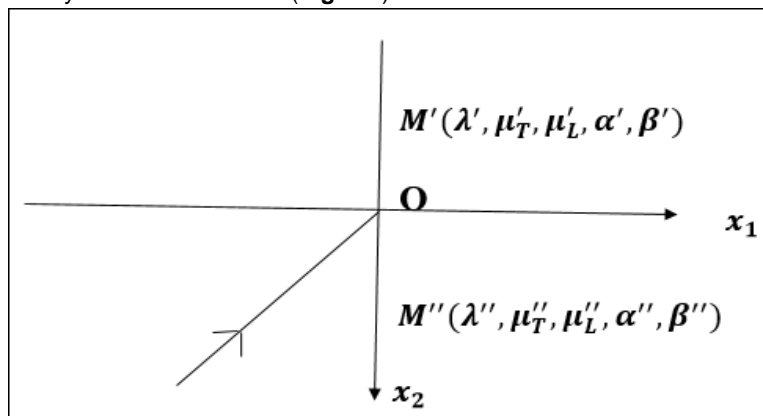


Fig. 1.3

The Stoneley wave's travels towards the mutual borderline of  $M'$  and  $M''$  in a fiber-reinforced elastic material.

The equation of motion for half space  $M'$  becomes

$$A'_{11} \frac{\partial^2 u'}{\partial x^2} + B'_2 \frac{\partial^2 v'}{\partial x \partial y} + B'_1 \frac{\partial^2 u'}{\partial y^2} = \rho' \frac{\partial^2 u'}{\partial t^2},$$

$$A'_{22} \frac{\partial^2 v'}{\partial x^2} + B'_2 \frac{\partial^2 u'}{\partial x \partial y} + B'_1 \frac{\partial^2 v'}{\partial y^2} = \rho' \frac{\partial^2 v'}{\partial t^2}. \dots\dots\dots (1.46)$$

Where  $\rho'$  is density and  $(A'_{11}, A'_{22}, B'_1, B'_2)$  are Fiber-reinforced parameters for the upper half space. Similarly, the equations for lower half space  $M''$  are

$$A''_{11} \frac{\partial^2 u''}{\partial x^2} + B''_2 \frac{\partial^2 v''}{\partial x \partial y} + B''_1 \frac{\partial^2 u''}{\partial y^2} = \rho'' \frac{\partial^2 u''}{\partial t^2},$$

$$A''_{22} \frac{\partial^2 v''}{\partial x^2} + B''_2 \frac{\partial^2 u''}{\partial x \partial y} + B''_1 \frac{\partial^2 v''}{\partial y^2} = \rho'' \frac{\partial^2 v''}{\partial t^2}. \dots\dots\dots (1.47)$$

Where  $\rho''$  is density and  $(A''_{11}, A''_{22}, B''_1, B''_2)$  refers to the quantities in lower half space.

We assume resolution for the equation (1.46) as

$$u' = (m'_1 V'_{21} e^{-q_1 k y} + m'_2 V'_{22} e^{-q_2 k y}) e^{ik(C_s t - x)},$$

$$v' = (V'_{21}e^{-q_1ky} + V'_{22}e^{-q_2ky})e^{ik(C_S t - x)}, \quad \dots\dots\dots (1.48)$$

Where  $m'_1 = iM'_1, m'_2 = iM'_2$ ,  $\dots\dots\dots (1.49)$

$$M'_1 = \frac{\rho' C_S^2 - B'_1 + A'_{22} q_1^2}{q_1 B'_2}, \dots\dots\dots (1.50)$$

and

$$M'_2 = \frac{\rho' C_S^2 - B'_1 + A'_{22} q_2^2}{q_2 B'_2}, \dots\dots\dots (1.51)$$

Where  $q'_1$  and  $q'_2$  are real and positive quantities defined in equatio?

$$q_1^2, q_2^2 = \frac{-\{(A'_{22} + B'_1)\rho' C_S^2 - A'_{11} A'_{22} - B_1'^2 + B_2'^2\} \pm \{[(A'_{22} + B'_1)\rho' C_S^2 - A'_{11} A'_{22} - B_1'^2 + B_2'^2]^2 - 4B'_1 A'_{22} (\rho' C_S^2 - A'_{11}) (\rho' C_S^2 - B_1')\}^{1/2}}{2B'_1 A'_{22}} \dots\dots\dots (1.52)$$

In  $M'$  medium the phase velocity of Stoneley wave's id denoted by  $C_S$ .

For lower medium  $m''_1$  and  $m''_2$  are defined as

$$m''_1 = iM''_1, m''_2 = iM''_2,$$

$$M''_1 = \frac{\rho'' C_S^2 - B''_1 + A''_{22} q_1^2}{q_1 B''_2},$$

And

$$M''_2 = \frac{\rho'' C_S^2 - B''_1 + A''_{22} q_2^2}{q_2 B''_2}, \dots\dots\dots (1.53)$$

Where  $q''_1$  and  $q''_2$  are real and positive quantity defined in equation (1.53) by changing dash to double dash.  $C_S$  is the phase velocity of stoneley waves in  $M''$  medium?

Similarly, the solution of (1.48) can be written as

$$u'' = (m''_1 V''_{21} e^{-q_1''ky} + m''_2 V''_{22} e^{-q_2''ky}) e^{ik(C_S t - x)},$$

$$v'' = (V''_{21} e^{-q_1''ky} + V''_{22} e^{-q_2''ky}) e^{ik(C_S t - x)}. \dots\dots\dots (1.54)$$

**Boundary Conditions**

The displacements in (1.48) and (1.54) must satisfy the boundary conditions,

$$\begin{aligned} \sigma'_{21} &= \sigma''_{21}, \\ \sigma'_{22} &= \sigma''_{22}, && \text{at } y=0 && \dots\dots\dots (1.55) \\ u' &= u'', \\ v' &= v'', \end{aligned}$$

Where  $\sigma'_{21} = \sigma'_L \left( \frac{\partial u'}{\partial y} + \frac{\partial v'}{\partial x} \right),$

$$\sigma'_{22} = A'_{12} \frac{\partial u'}{\partial y} + A'_{22} \frac{\partial v'}{\partial y}, \dots\dots\dots (1.56)$$

$$\sigma''_{21} = \mu''_L \left( \frac{\partial u''}{\partial y} + \frac{\partial v''}{\partial x} \right),$$

$$\sigma''_{22} = A''_{12} \frac{\partial u''}{\partial y} + A''_{22} \frac{\partial v''}{\partial y},$$

Using equations (1.48) and (1.54) in equation (1.51), we get

$$\mu'_L (m'_1 q'_1 + i) V'_{21} + \mu'_L (m'_2 q'_2 + i) V'_{22} - V'_{21} \mu''_L (m''_1 q''_1 + i) - \mu''_L (m''_2 q''_2 + i) V'_{22} = 0, \dots\dots\dots (1.57)$$

$$(A'_{12} m'_1 i + A'_{22} q'_1) V'_{21} + (m'_2 A'_{12} i + A'_{22} q'_2) V'_{22} - (A''_{12} m''_1 i + A''_{22} q''_1) V'_{21} - (m''_2 A''_{12} i + A''_{22} q''_2) V'_{22} = 0, \dots\dots\dots (1.58)$$

$$m'_1 V'_{21} + m'_2 V'_{22} - m''_1 V'_{21} - m''_2 V'_{22} = 0 \dots\dots\dots (1.59)$$

$$V'_{21} + V'_{22} - V''_{21} - V''_{22} = 0, \dots\dots\dots (1.60)$$

To conclude we acquired the expression for the velocity of the wave in the mutual borderline of medium  $M'$  and  $M''$  by excluding  $V'_{21}, V'_{22}, V''_{21}$  and  $V''_{22}$  from equations (1.57)-(1.60) as

$$\begin{aligned} &|11 - 1 - 1m'_1 m'_2 - m''_1 - m''_2 (m'_1 q'_1 + i) \mu'_L (m'_2 q'_2 + i) \mu'_L (-m''_1 q''_1 + i) \mu''_L (-m''_2 q''_2 + i) \mu''_L (A'_{12} m'_1 i + A'_{22} q'_1) (m'_2 A'_{12} i + A'_{22} q'_2) - (A''_{12} m''_1 i \\ &+ A''_{22} q''_1) - (m''_2 A''_{12} i + A''_{22} q''_2)| \\ &= 0 \dots\dots\dots (1.61) \end{aligned}$$

**PARTICULAR CASE**

If we put  $\alpha', \beta'$  and  $|\mu'_L - \mu'_T|$  equal to zero, the elastic co-fficients for medium  $M'$  become

$$A'_{11} = \lambda' + 2\mu'_T, A'_{12} = \lambda', \quad B_1 = \mu'_T,$$

$$A'_{22} = \lambda' + 2\mu'_T, A'_{23} = \lambda', B_2 = \lambda' + \mu'_T \dots\dots\dots(1.62)$$

From equations (1.50) and (1.51), we get

$$M'_1 = q'_1, M'_2 = \frac{1}{q'_2}, \dots\dots\dots(1.63)$$

$$q'^2_1 = \left(1 - \frac{c_s'^2}{\beta'^2}\right), q'^2_2 = \left(1 - \frac{c_s'^2}{\alpha'^2}\right) \dots\dots\dots(1.64)$$

where

$$\alpha'^2 = \frac{\lambda' + 2\mu'_T}{\rho'}, \text{ And } \beta'^2 = \frac{\mu'_T}{\rho'} \dots\dots\dots (1.65)$$

$\lambda', \mu'_T$  Are Lamé constants for upper half-space  $M'$  ? Similarly the quantities defined in Equations (1.62)-(1.65) by changing dash to double dash show for lower half space. Putting These values in equation (1.61), we acquire the Stoneley wave velocity equation in homogenous isotropic elastic half-spaces.

**1.7 Conclusion**

It has been observed that in a Fiber-reinforced material the propagation of surface waves is influenced by the reinforced parameters. It is clear from equations (1.34) and (1.38) for the presence of Love waves that  $p_i$  should be imaginary and  $p'_1$  is real satisfied if

$$\left(\frac{\mu'_L}{\rho'}\right) < C_L^2 < \left(\frac{\mu_L}{\rho}\right) \dots\dots\dots (1.66)$$

Equation (1.66) depends upon the reinforced parameters  $\mu_L$  and  $\mu'_L$ .

At the end we reached to a conclusion that the velocity of the Rayleigh waves is comparatively higher in a Fiber-reinforced elastic medium as compared to that of isotropic media. It is clear from equation (1.66) that a Stoneley wave is affected by the reinforced parameters.

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